N9924B



Fact Book

INTRODUCTION

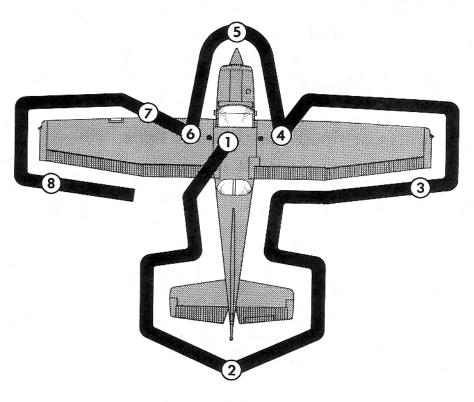
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Section 4 provides checklist and amplified procedures for the conduct of normal operation. Normal procedures associated with optional systems can be found in Section 9.

SPEEDS FOR NORMAL OPERATION

Unless otherwise noted, the following speeds are based on a maximum weight of 2650 pounds and may be used for any lesser weight. However, to achieve the performance specified in Section 5 for takeoff distance, the speed appropriate to the particular weight must be used.

| Takeoff: |
|---|
| Normal Climb Out |
| Short Field Takeoff, Flaps 0°, Speed at 50 Feet 63 KIAS |
| Enroute Climb, Flaps and Gear Up: |
| Normal |
| Best Rate of Climb, Sea Level 84 KIAS |
| Best Rate of Climb, 10,000 Feet |
| Best Angle of Climb, Sea Level 67 KIAS |
| Best Angle of Climb, 10,000 Feet |
| Landing Approach: |
| Normal Approach, Flaps Up 70-80 KIAS |
| Normal Approach, Flaps 30° 60-70 KIAS |
| Short Field Approach, Flaps 30° 63 KIAS |
| Balked Landing: |
| Maximum Power, Flaps 20° |
| Maximum Recommended Turbulent Air Penetration Speed: |
| 2650 Lbs |
| 2250 Lbs |
| 1850 Lbs |
| Maximum Demonstrated Crosswind Velocity: |
| Takeoff or Landing |
| |



NOTE

Visually check airplane for general condition during walk-around inspection. Use of the refueling steps and assist handles (if installed) will simplify access to the upper wing surfaces for visual checks and refueling operations. In cold weather, remove even small accumulations of frost, ice or snow from wing, tail and control surfaces. Also, make sure that control surfaces contain no internal accumulations of ice or debris. Prior to flight, check that pitot heater (if installed) is warm to touch within 30 seconds with battery and pitot heat switches on. If a night flight is planned, check operation of all lights, and make sure a flashlight is available.

Figure 4-1. Preflight Inspection

CHECKLIST PROCEDURES

PREFLIGHT INSPECTION

- 1 CABIN
 - 1. Pilot's Operating Handbook -- AVAILABLE IN THE AIRPLANE.
 - 2. Parking Brake -- SET
 - Control Wheel Lock -- REMOVE.
 Avionics Power Switch -- OFF.
 - 5. Ignition Switch -- OFF.
 - 6. Landing Gear Lever -- DOWN.
 - 7. Master Switch -- ON.

WARNING

When turning on the master switch, using an external power source, or pulling the propeller through by hand, treat the propeller as if the ignition switch were on. Do not stand, nor allow anyone else to stand, within the arc of the propeller, since a loose or broken wire, or a component malfunction, could cause the propeller to rotate.

- 8. Fuel Quantity Indicators -- CHECK QUANTITY.
- 9. Landing Gear Position Indicator Light (green) -- ILLUMINATED
- 10. Avionics Cooling Fan -- CHECK AUDIBLY FOR OPERATION.
- Master Switch -- OFF.
- 12. Static Pressure Alternate Source Valve (if installed) -- OFF.
- 13. Fuel Selector Valve -- BOTH.
- 14. Baggage Door -- CHECK for security.

(2) EMPENNAGE

- Rudder Gust Lock -- REMOVE.
- 2. Tail Tie-Down -- DISCONNECT.
- 3. Control Surfaces -- CHECK freedom of movement and security.

(3) RIGHT WING Trailing Edge

Aileron -- CHECK freedom of movement and security.

4 RIGHT WING

Wing Tie-Down -- DISCONNECT.

- 2. Main Wheel Tire and Wheel Well -- CHECK tire for proper inflation and wheel well for condition and cleanliness.
- Fuel Tank Sump Quick-Drain Valve -- DRAIN fuel (using sampler cup) to check for water, sediment, and proper fuel grade before first flight of day and after each refueling. If water is observed, take further samples until there is no evidence of water contamination.
- 4. Fuel Selector Quick-Drain Valve (on bottom of fuselage) -- DRAIN fuel (using sampler cup) to check for water, sediment, and proper fuel grade before first flight of day and after each refueling. If water is observed, take further samples until there is no evidence of water contamination.
- 5. Fuel Quantity -- CHECK VISUALLY for desired level.
- 6. Fuel Filler Cap -- SECURE and vent unobstructed.

5 NOSE

- Static Source Openings (both sides of fuselage) -- CHECK for stoppage.
- 2. Engine Oil Dipstick/Filler Cap -- CHECK oil level, then check dipstick/filler cap SECURE. Do not operate with less than five quarts. Fill to eight quarts for extended flight.
- Propeller and Spinner -- CHECK for nicks and security, and oil leaks.
- 4. Nose Gear Doors -- CHECK for security.
- Nose Wheel Tire, Strut and Wheel Well -- CHECK tire and strut for proper inflation and wheel well for condition and cleanliness.
- 6. Fuel Strainer Quick-Drain Valve (on lower left side of engine cowl)
 -- DRAIN fuel (using sampler cup) to clear strainer of possible
 water and sediment before first flight of day and after each
 refueling. If water is observed, the fuel system may contain
 additional water, and further draining of the system at the strainer,
 fuel tank sumps, and fuel selector quick-drain valve must be
 accomplished.
- Nose Tie-Down -- DISCONNECT.

6 LEFT WING

- 1. Main Wheel Tire and Wheel Well -- CHECK tire for proper inflation and wheel well for condition and cleanliness.
- Fuel Tank Sump Quick-Drain Valve -- DRAIN fuel (using sampler cup) to check for water, sediment, and proper fuel grade before first flight of day and after each refueling. If water is observed, take further samples until there is no evidence of water contamination.
- 3. Fuel Quantity -- CHECK VISUALLY for desired level.
- Fuel Filler Cap -- SECURE.

SECTION

LEFT WING Leading Edge 1. Pitot Tube Cover -- REMOVE and check opening for stoppage.

Fuel Tank Vent Opening -- CHECK for stoppage. 2.

Aileron -- CHECK for freedom of movement and security.

- Stall Warning Vane -- CHECK for freedom of movement whil 3.
- master switch is momentarily turned ON (horn should sound when vane is pushed upward).
- Wing Tie-Down -- DISCONNECT. 4.
- Landing Lights -- CHECK for condition and cleanliness of cover 5.
- LEFT WING Trailing Edge

1. Preflight Inspection -- COMPLETE.

BEFORE STARTING ENGINE

- 2. Passenger Briefing -- COMPLETE.
 - 3. Seats, Seat Belts, Shoulder Harnesses -- ADJUST and LOCK.
- 4. Brakes -- TEST and SET.
- Avionics Power Switch -- OFF. 5.

CAUTION

The avionics power switch must be OFF during engine

start to prevent possible damage to avionics.

- 6. Circuit Breakers -- CHECK IN.
- 7. Electrical Equipment, Autopilot (if installed) -- OFF.
- 8. Landing Gear Lever -- DOWN
- Cowl Flaps -- OPEN (move lever out of locking hole to reposition) 9. 10. Fuel Selector Valve -- BOTH.

STARTING ENGINE

- 1. Carburetor Heat -- COLD. 2.
- Propeller -- HIGH RPM. Mixture -- RICH. 3.
- Propeller Area -- CLEAR. 4.
- 5. Master Switch -- ON.
- Throttle -- PUMP once or twice; leave open 1/4 inch. If engine is 6.

hot, turn auxiliary fuel pump ON during start.

- 7. Ignition Switch -- START (release when engine starts).
- Oil Pressure -- CHECK.
- 9. Avionics Power Switch -- ON.
- 10. Navigation Lights and Flashing Beacon -- ON as required.
- 11. Radios -- ON.

BEFORE TAKEOFF

- 1. Parking Brake -- SET.
- 2. Seats, Seat Belts, Shoulder Harnesses -- CHECK SECURE.
- 3. Cabin Doors -- CLOSED and LOCKED.
- 4. Flight Controls -- FREE and CORRECT.
- 5. Flight Instruments -- CHECK and SET.
- 6. Fuel Quantity -- CHECK.
- Auxilary Fuel Pump -- ON (check for rise in fuel pressure), then OFF.

NOTE

In flight, gravity feed will normally supply satisfactory fuel flow if the engine-driven fuel pump should fail. However, if a fuel pump failure causes the fuel pressure to drop below 0.5 PSI, use the auxiliary fuel pump to assure proper engine operation.

- 8. Mixture -- RICH.
- 9. Fuel Selector Valve -- RECHECK BOTH.
- 10. Elevator and Rudder Trim -- SET for takeoff.
- 11. Throttle -- 1800 RPM.
 - Magnetos -- CHECK (RPM drop should not exceed 150 RPM or either magneto or 50 RPM differential between magnetos. I RPM drop is excessive, lean to smooth operation and recheck)
 - b. Carburetor Heat -- CHECK (for RPM drop).
 - c. Propeller -- CYCLE from high to low RPM; return to high RPM (full in).
 - d. Mixture -- RICH (below 3000 feet).
 - e. Suction Gage -- CHECK.
 - f. Engine Instruments and Ammeter -- CHECK.
- 12. Throttle -- 1000 RPM or less.
- 13. Throttle Friction Lock -- ADJUST.
- 14. Strobe Lights (if installed) -- AS DESIRED.
- 15. Radios and Avionics -- SET.
- 16. Autopilot (if installed) -- OFF.
- 17. Wing Flaps -- SET for takeoff (see Takeoff checklists).
- 18. Parking Brake -- RELEASE.

SECTION

TAKEOFF

NORMAL TAKEOFF

- Wing Flaps -- 0°.
 - Carburetor Heat -- COLD.
 Power -- FULL THROTTI.
 - Power -- FULL THROTTLE and 2700 RPM.
 Mixture -- RICH (mixture may be leaned above 3000 feet to obtain
 - maximum power).

 5. Elevator Control -- LIFT NOSE WHEEL at 55 KIAS.

NOTE

When the nose wheel is lifted, the gear motor may run 1-2 seconds to restore hydraulic pressure.

- 6. Climb Speed -- 70-80 KIAS.7. Brakes -- APPLY momentarily when airborne.
- 7. Brakes -- APPLY momentarily when airborne 8. Landing Gear -- RETRACT in climb out.

SHORT FIELD TAKEOFF

- 1. Wing Flaps -- 0°.
- 2. Carburetor Heat -- COLD.
- Brakes -- APPLY.
- 4. Power -- FULL THROTTLE and 2700 RPM.
- 5. Mixture -- RICH (LEAN above 3000 feet to obtain maximum
- power).
- 6. Brakes -- RELEASE.
- Elevator Control -- MAINTAIN SLIGHTLY TAIL-LOW ATTI TUDE.
- 8. Climb Speed -- 63 KIAS until all obstacles are cleared.
- 9. Landing Gear -- RETRACT after obstacles are cleared.

ENROUTE CLIMB

NORMAL CLIMB

- Airspeed -- 85-95 KIAS.
 Power -- 25 INCHES Hg and 2500 R.
 - 2. Power -- 25 INCHES Hg and 2500 RPM.
- Fuel Selector Valve -- BOTH.
 Mixture -- FULL RICH (mixture may be leaned above 3000 feet)
- 4. Mixture -- FULL RICH (mixture may be leaned above 3000 feet 5. Cowl Flaps -- OPEN as required.

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MAXIMUM PERFORMANCE CLIMB

- 1. Airspeed -- 84 KIAS at sea level to 77 KIAS at 10,000 feet.
- 2. Power -- FULL THROTTLE and 2700 RPM.
- Fuel Selector Valve -- BOTH.
- 4. Mixture -- FULL RICH (mixture may be leaned above 3000 feet).
- 5. Cowl Flaps -- FULL OPEN.

CRUISE

- 1. Power -- 15-25 INCHES Hg, 2100-2700 RPM.
- 2. Elevator and Rudder Trim -- ADJUST.
- Mixture -- LEAN.
- 4. Cowl Flaps -- CLOSED.

DESCENT

- Fuel Selector Valve -- BOTH.
- Power -- AS DESIRED.
- 3. Mixture -- ENRICHEN as required.
- Carburetor Heat -- FULL HEAT AS REQUIRED to prevent carburetor icing.
- 5. Cowl Flaps -- CLOSED.
- Wing Flaps -- AS DESIRED (0° 10° below 130 KIAS, 10° 30° below 100 KIAS).

NOTE

The landing gear may be extended below 140 KIAS to increase the rate of descent.

BEFORE LANDING

- Seats, Seat Belts, Shoulder Harnesses -- SECURE.
- 2. Fuel Selector Valve -- BOTH.
- 3. Landing Gear -- DOWN (below 140 KIAS).
- 4. Landing Gear -- CHECK (observe main gear down and green indicator light illuminated).
- Mixture -- RICH.
- 6. Propeller -- HIGH RPM.
- 7. Carburetor Heat -- ON (apply full heat before closing throttle).
- 8. Autopilot (if installed) -- OFF.

TAKEOFF

NORMAL TAKEOFF

- 1. Wing Flaps -- 0°.
- Carburetor Heat -- COLD.
- Power -- FULL THROTTLE and 2700 RPM.
- Mixture -- RICH (mixture may be leaned above 3000 feet to obtain maximum power).
- 5. Elevator Control -- LIFT NOSE WHEEL at 55 KIAS.

NOTE

When the nose wheel is lifted, the gear motor may run 1-2 seconds to restore hydraulic pressure.

- 6. Climb Speed -- 70-80 KIAS. 7
- 7. Brakes -- APPLY momentarily when airborne.
- 8. Landing Gear -- RETRACT in climb out.

SHORT FIELD TAKEOFF

- Wing Flaps -- 0°.
- 2. Carburetor Heat -- COLD.
- 3. Brakes -- APPLY.
- Power -- FULL THROTTLE and 2700 RPM.
- 5. Mixture -- RICH (LEAN above 3000 feet to obtain maximum power).
- 6. Brakes -- RELEASE.
- 7. Elevator Control -- MAINTAIN SLIGHTLY TAIL-LOW ATTITUDE.
- 8. Climb Speed -- 63 KIAS until all obstacles are cleared.
- 9. Landing Gear -- RETRACT after obstacles are cleared.

ENROUTE CLIMB

NORMAL CLIMB

- Airspeed -- 85-95 KIAS.
- 2. Power -- 25 INCHES Hg and 2500 RPM.
- Fuel Selector Valve -- BOTH.
- 4. Mixture -- FULL RICH (mixture may be leaned above 3000 feet).
- 5. Cowl Flaps -- OPEN as required.

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SECTION 4

LANDING

NORMAL LANDING

- Airspeed (70/80 KIAS (flaps UP). 1. Wing Flaps -- AS DESIRED (0°- 10° below 130 KIAS, 10°-30° below 2.
- 100 KIAS).
- 3. Airspeed -- 60-70 KIAS (flaps DOWN).
- 4. Trim -- ADJUST.
- 5. Touchdown -- MAIN WHEELS FIRST.
- 6. Landing Roll -- LOWER NOSE WHEEL GENTLY. Braking -- MINIMUM REQUIRED. 7.

SHORT FIELD LANDING

8.

- Airspeed -- 70-80 KIAS (flaps UP). 1. Wing Flaps -- 30° (below 100 KIAS). 2.
 - 3. Airspeed -- MAINTAIN 63 KIAS.
 - 4. Trim -- ADJUST.
 - Power -- REDUCE to idle as obstacle is cleared. 5.

Wing Flaps -- RETRACT for maximum brake effectiveness.

- 6. Touchdown -- MAIN WHEELS FIRST. 7. Brakes -- APPLY HEAVILY.
- BALKED LANDING

1. Power -- FULL THROTTLE and 2700 RPM. 2. Carburetor Heat -- COLD.

- Wing Flaps -- RETRACT to 20°.
- Climb Speed -- 55 KIAS. 4.
- 5. Wing Flaps -- RETRACT slowly after reaching 65 KIAS.

AFTER LANDING

Cowl Flaps -- OPEN.

- 1. Carburetor Heat -- COLD.
- Wing Flaps -- UP. Cowl Flaps -- OPEN.

SECURING AIRPLANE

- 1. Parking Brake -- SET.
- Throttle -- 1000 RPM.

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- 3. Avionics Power Switch -- OFF.
- 4. Electrical Equipment -- OFF.
- Mixture -- IDLE CUT-OFF (pulled full out). Throttle -- CLOSE as RPM drops. 5. 6.
- 7. Ignition Switch -- OFF. Master Switch -- OFF. 8.
- Control Lock -- INSTALL. 9.
- Cowl Flaps -- CLOSE. 10.

AIRSPEED CALIBRATION

NORMAL STATIC SOURCE

CONDITIONS:

Power required for level flight or maximum power during descent.

| FLAPS UP | | | | | | | | | | |
|-----------------------|----------|----------|----------|----------|----------|-----------|------------|------------|--------------------|--------------------|
| KIAS KCAS | 50 55 | 60 63 | 70 71 | 80 80 | 90 89 | 100 99 | 110 108 | 120 118 | 130 140 128 138 | 150 160 147 157 |
| FLAPS 10 ^o | | | | | | | | | | |
| KIAS KCAS | 40 50 | 50 54 | 60 62 | 70 71 | 80 81 | 90 91 | 100 100 | 110 110 | 120 130 120 130 | |
| FLAPS 30° | | | | , | | | | | | |
| KIAS KCAS | 40 47 | 50 54 | 60 62 | 70 71 | 80 81 | 90 90 | 100 101 | | | |

Figure 5-1. Airspeed Calibration (Sheet 1 of 2)

AIRSPEED CALIBRATION ALTERNATE STATIC SOURCE

HEATER AND DEFROSTER FULL ON AND WINDOWS CLOSED

| FLAPS UP | | | | | | | | | | |
|-----------------------------------|----------|----------|----------|----------|----------|------------|------------|------------|------------|------------|
| NORMAL KIAS ALTERNATE KIAS | 50 46 | 60 58 | 70 69 | 80 79 | 90 88 | 100 98 | 110 108 | 120 117 | 140 136 | 160 155 |
| FLAPS 10 ⁰ | | | | | | | | | | |
| NORMAL KIAS ALTERNATE KIAS | 50 45 | 60 58 | 70 69 | 80 80 | 90 90 | 100 100 | 110 109 | 120 119 | 130 128 | |
| FLAPS 30° | | | | | | | | | | |
| NORMAL KIAS ALTERNATE KIAS | 50 46 | 60 58 | 70 68 | 80 78 | 90 87 | 100 96 | | | | |

Figure 5-1. Airspeed Calibration (Sheet 2 of 2)

STALL SPEEDS

CONDITIONS: Power Off

Gear Up or Down

NOTES:

- 1. Altitude loss during a stall recovery may be as much as 230 feet.
- 2. KIAS values are approximate.

MOST REARWARD CENTER OF GRAVITY

| | | ANGLE OF BANK | | | | | | | | | |
|---------------|--------------------|---------------|------|------|------|------|----------------|------|------|--|--|
| WEIGHT LBS | FLAP DEFLECTION | C |)o | 3 | 0° | 4 | 5 ⁰ | 6 | 0° | | |
| | | KIAS | KCAS | KIAS | KCAS | KIAS | KCAS | KIAS | KCAS | | |
| | UP | 46 | 54 | 49 | 58 | 55 | 64 | 65 | 76 | | |
| 2650 | 10 ⁰ | 42 | 52 | 45 | 56 | 50 | 62 | 59 | 74 | | |
| | 30° | 39 | 50 | 42 | 54 | 47 | 59 | 56 | 71 | | |

MOST FORWARD CENTER OF GRAVITY

| | | | | | A | NGLE (| OF BAN | K | | |
|---|---------------|--------------------|------|------|------|--------|-----------------|------|------|------|
| | WEIGHT LBS | FLAP DEFLECTION | 0° | | 30° | | 45 ⁰ | | 60° | |
| | | | KIAS | KCAS | KIAS | KCAS | KIAS | KCAS | KIAS | KCAS |
| | | UP | 50 | 57 | 54 | 61 | 59 | 68 | 71 | 81 |
| | 2650 | 10 ⁰ | 46 | 54 | 49 | 58 | 55 | 64 | 65 | 76 |
| - | | 30° | 42 | 51 | 45 | 55 | 50 | 61 | 59 | 72 |

Figure 5-3. Stall Speeds

MAXIMUM WEIGHT 2650 LBS

CONDITIONS:

Paved, Level Dry Runway

Flaps Up

2700 RPM and Full Throttle Prior to Brake Release Cowl Flaps Open

SHORT FIELD

NOTES:

Zero Wind

- 1. Short field technique as specified in Section 4.
- Prior to takeoff from fields above 3000 feet elevation, the mixture should be leaned to give maximum power in a full throttle, static runup.
- 3. Decrease distances 10% for each 9 knots headwind. For operation with tailwinds up to 10 knots, increase distances by 10% for each 2 knots.
- 4. For operation on a dry, grass runway, increase distances by 15% of the "ground roll" figure.

| | T | | PEED PRESS | | ESS 0°C | | 10 ⁰ C | | 20°C | | 30°C | | 40°C | |
|---------------|------|-------------------|--|---|--|--|--|--|--|--|--|--|--|--|
| WEIGHT LBS | LIFT | AS AT 50 FT | | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | |
| 2650 | 58 | 63 | S.L. 1000 2000 3000 4000 5000 6000 7000 8000 | 955 1040 1140 1250 1370 1505 1660 1830 2025 | 1605 1755 1925 2120 2345 2600 2905 3265 3700 | 1025 1120 1225 1340 1475 1620 1785 1970 2180 | 1715 1880 2065 2280 2525 2805 3145 3545 4045 | 1100 1200 1315 1440 1585 1740 1920 2120 2350 | 1835 2015 2215 2450 2715 3030 3405 3865 4440 | 1175 1285 1410 1545 1700 1870 2065 2280 2530 | 1965 2155 2380 2630 2930 3280 3700 4220 4895 | 1260 1375 1510 1655 1820 2005 2215 2450 2720 | 2100 2310 2550 2830 3160 3550 4025 4630 5430 | |

MODEL 172R

TAKEOFF DISTANCE

2500 LBS AND 2300 LBS

SHORT FIELD

REFER TO SHEET 1 FOR APPROPRIATE CONDITIONS AND NOTES.

| | | TAKEOFF SPEED KIAS | | | 0°C | | 10°C | | 20°C | ; | 30°C | | 40 ^о С |
|---------------|----|--------------------------|--|---|--|--|--|---|--|--|--|--|--|
| WEIGHT LBS | | AS | PRESS ALT FT | | | | TOTAL TO CLEAR 50 FT OBS | | TOTAL TO CLEAR 50 FT OBS | | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS |
| 2500 | 56 | 61 | S.L. 1000 2000 3000 4000 5000 6000 7000 8000 | 835 910 995 1090 1195 1310 1440 1585 1755 | 1400 1525 1670 1835 2015 2230 2470 2760 3095 | 895 975 1070 1170 1280 1410 1550 1710 1890 | 1495 1635 1790 1965 2165 2400 2665 2980 3360 | 960 1045 1145 1255 1375 1515 1665 1840 2035 | 1595 1745 1915 2105 2325 2580 2875 3230 3655 | 1025 1120 1225 1345 1475 1625 1790 1975 2185 | 1705 1865 2050 2260 2500 2780 3105 3500 3980 | 1100 1200 1315 1440 1580 1740 1920 2120 2350 | 1820 1995 2195 2420 2685 2990 3355 3800 4350 |
| 2300 | 54 | 59 | S.L. 1000 2000 3000 4000 5000 6000 7000 8000 | 690 750 820 895 980 1075 1180 1295 1430 | 1160 1265 1380 1505 1650 1815 2005 2220 2465 | 740 805 880 960 1050 1155 1265 1395 1540 | 1240 1350 1475 1610 1770 1950 2150 2385 2660 | 790 860 940 1030 1130 1240 1360 1500 1655 | 1320 1440 1575 1725 1895 2090 2310 2570 2875 | 845 920 1010 1105 1210 1325 1460 1610 1775 | 1405 1535 1680 1845 2025 2240 2485 2765 3105 | 905 985 1080 1180 1295 1420 1565 1725 1905 | 1500 1635 1795 1970 2170 2400 2670 2980 3355 |

Timere E E Telegraff Distance (Chest 9 of 9)

MAXIMUM RATE OF CLIMB

Flaps Up Gear Up 2700 RPM Full Throttle

Mixture Leaned above 3000 Feet

Cowl Flaps Open

CONDITIONS:

| 2 2 4 4 | | | | | | |
|---------|--|--|---|---|---|---------------------------------|
| WEIGHT | PRESS ALT | CLIMB SPEED | | RATE OF C | LIMB - FPM | |
| LBS | FT | KIAS | -20 ^o C | 0°C | 20°C | 40°C |
| 2650 | S.L. 2000 4000 6000 8000 10,000 12,000 | 84 83 81 80 78 77 75 | 925 825 720 620 525 430 330 | 855 755 655 560 465 370 275 | 780 685 590 495 405 310 220 | 710 620 525 435 340 |

Figure 5-6. Maximum Rate of Climb

TIME, FUEL, AND DISTANCE TO CLIMB

MAXIMUM RATE OF CLIMB

CONDITIONS:

Cowl Flaps Open Standard Temperature

Flaps Up Gear Up 2700 RPM Full Throttle Mixture Leaned above 3000 Feet

NOTES:

- 1. Add 1.4 gallons of fuel for engine start, taxi, and takeoff allowance.
- 2. Increase time, fuel and distance by 10% for each 10°C above standard temperature.
- 3. Distances shown are based on zero wind.

| WEIGHT | PRESSURE | TEMP | CLIMB | RATE OF | F | ROM SEA LE | VEL |
|--------|----------------|------|---------------|--------------|-------------|----------------------|----------------|
| LBS | ALTITUDE FT | °C | SPEED KIAS | CLIMB FPM | TIME MIN | FUEL USED GALLONS | DISTANCE NM |
| 2650 | S.L. | 15 | 84 | 800 | 0 | 0.0 | 0 |
| | 1000 | 13 | 83 | 760 | 1 | 0.4 | 2 |
| | 2000 | 11 | 83 | 715 | 3 | 0.8 | 4 |
| | 3000 | 9 | 82 | 675 | 4 | 1.1 | 6 |
| | 4000 | 7. | 81 | 635 | 6 | 1.6 | 8 |
| | 5000 | 5 | 81 | 590 | 7 | 2.0 | 10 |
| | 6000 | 3 | 80 | 550 | 9 | 2.4 | 13 |
| | 7000 | 1 | 79 | 510 | 11 | 2.9 | 16 |
| yA. | 8000 | -1 | 78 | 465 | 13 | 3.3 | 19 |
| | 9000 | -3 | 78 | 425 | 15 | 3.8 | 22 |
| | 10,000 | -5 | 77 | 385 | 18 | 4.3 | 26 |
| | 11,000 | -7 | 76 | 340 | 21 | 4.9 | 30 |
| | 12,000 | -9 | 75 | 300 | 24 | 5.5 | 35 |
| | | | | | | | |
| | | ı | | 1 | | | ı |

Figure 5-7. Time, Fuel, and Distance to Climb (Sheet 1 of 2)

PRESSURE ALTITUDE 4000 FEET

CONDITIONS: 2650 Pounds Recommended Lean Mixture

Cowl Flaps Closed

NOTE
For best fuel economy, operate at the leanest mixture that results in smooth engine operation or at peak EGT if an EGT indicator is installed.

| | | 1 | OC BELC NDARD 1 -13 ^O C | | | TANDAR IPERATU 7 ^O C | | | °C ABOV NDARD T 27°C | |
|------|----------------------------|----------------------------|--|---------------------------------|----------------------------|---------------------------------------|---------------------------------|----------------------------|--------------------------------|---------------------------------|
| RPM | MP | % BHP | KTAS | GPH | % BHP | KTAS | GPH | % BHP | KTAS | GPH |
| 2500 | 24 23 22 21 | 75 70 66 | 131 127 124 | 10.0 9.4 8.8 | 77 72 68 63 | 135 132 128 124 | 10.2 9.7 9.1 8.6 | 74 70 66 61 | 136 132 128 124 | 9.9 9.4 8.8 8.3 |
| 2400 | 24 23 22 21 | 77 72 68 63 | 132 129 126 122 | 10.2 9.7 9.1 8.6 | 74 70 65 61 | 133 130 126 121 | 9.9 9.3 8.8 8.3 | 72 67 63 59 | 134 130 126 121 | 9.6 9.0 8.5 8.1 |
| 2300 | 25 24 23 22 | 74 70 65 | 130 127 123 | 9.9 9.3 8.8 | 76 71 67 63 | 134 131 127 123 | 10.1 9.5 9.0 8.5 | 73 69 65 61 | 135 131 127 123 | 9.7 9.2 8.7 8.3 |
| 2200 | 24 23 22 21 | 71 67 63 58 | 128 125 121 116 | 9.5 9.0 8.5 8.0 | 69 65 60 56 | 129 125 121 116 | 9,2 8.7 8.2 7.7 | 66 62 58 54 | 129 125 120 115 | 8.9 8.4 8.0 7.5 |
| 2100 | 23 22 21 20 19 | 64 60 56 52 48 | 122 118 114 109 103 | 8.6 8.2 7.7 7.2 6.7 | 62 58 54 50 46 | 122 118 113 108 101 | 8.4 7.9 7.4 7.0 6.6 | 60 56 52 48 44 | 122 117 112 106 98 | 8.1 7.7 7.2 6.8 6.4 |

Figure 5-8. Cruise Performance (Sheet 2 of 6)

PRESSURE ALTITUDE 6000 FEET

CONDITIONS: 2650 Pounds

Recommended Lean Mixture Cowl Flaps Closed NOTE
For best fuel economy, operate at the leanest mixture that results in smooth engine operation or at peak EGT if an EGT indicator is installed.

9.0

8.5

8.0

8.7

8.2

7.7

7.3

6.8

| | | | OC BELO NDARD 1 -17 ^O C | | | TANDAR IPERATU 3 ^O C | | | °C ABOV NDARD T 23°C | | |
|------|----------------------|----------------------|--|---------------------------|----------------------|---------------------------------------|---------------------------|----------------------|----------------------------|--------------------------|--|
| RPM | MP | % BHP | KTAS | GPH | % BHP | KTAS | GPH | % BHP | KTAS | GPH | |
| 2500 | 23 22 21 20 | 73 68 63 | 132 128 123 | 9.7 9.1 8.6 | 75 70 66 61 | 136 132 128 123 | 10.0 9.4 8.8 8.3 | 72 68 63 59 | 136 132 128 123 | 9.6 9.1 8.6 8.1 | |
| 2400 | 24 23 22 21 | 75 70 66 | 133 130 126 | 10.0 9.4 8.8 | 77 72 68 63 | 137 134 130 126 | 10.2 9.6 9.1 8.6 | 74 70 66 61 | 138 134 130 125 | 9.9 9.3 8.8 8.3 | |
| 2300 | 24 23 22 21 | 77 72 68 63 | 134 131 127 123 | 10.2 9.6 9.1 8.5 | 74 70 65 61 | 135 132 128 123 | 9.8 9.3 8.8 8.3 | 71 67 63 59 | 136 132 127 123 | 9.5 9.0 8.5 8.0 | |
| 2200 | 24 | 74 | 132 | 9.9 | 71 | 133 | 9.5 | 69 | 133 | 9.2 | |

Figure 5-8. Cruise Performance (Sheet 3 of 6)

9.3

8.8

8.3

8.9

8.5

8.0

7.5

7.0

8.7

8.2

7.8

8.4

7.9

7.5

7.0

6.6

PRESSURE ALTITUDE 8000 FEET

CONDITIONS: 2650 Pounds Recommended Lean Mixture Cowl Flaps Closed

NOTE
For best fuel economy, operate at the leanest mixture that results in smooth engine operation or at peak EGT if an EGT indicator is installed.

| | | | °C BELO NDARD 1 -21°C | | _ | TANDAR IPERATU - 1 ^O C | - | | OC ABOV NDARD T 19 ^O C | | |
|------|----------------|----------|-----------------------------|------------|----------------|---|--------------------|----------------|---|-------------------|--|
| RPM | MP | % BHP | KTAS | GPH | % BHP | KTAS | GPH | % BHP | KTAS | GPH | |
| 2700 | 22 21 | 75 | 135 | 10.0 | 77 72 | 140 136 | 10.3 9.6 | 74 70 | 141 136 | 9.9 9.3 | |
| 2600 | 22 21 20 | 73 68 | 134 129 | 9.7 9.1 | 75 70 65 | 138 134 129 | 10.0 9.4 8.8 | 72 68 63 | 139 134 129 | 9.6 9.1 8.5 | |
| 2500 | 22 | 75 | 136 | 10.0 | 73 | 136 | 9.7 | 70 | 137 | 9.4 | |
| | 21 | 71 | 132 | 9.4 | 68 | 132 | 9.1 | 66 | 132 | 8.8 | |
| | 20 | 66 | 128 | 8.8 | 63 | 127 | 8.6 | 61 | 127 | 8.3 | |
| | 19 | 61 | 123 | 8.3 | 59 | 122 | 8.0 | 57 | 121 | 7.8 | |
| 2400 | 22 | 73 | 134 | 9.7 | 70 | 134 | 9.4 | 68 | 134 | 9.1 | |
| | 21 | 68 | 130 | 9.1 | 66 | 130 | 8.8 | 63 | 129 | 8.6 | |
| | 20 | 64 | 125 | 8.6 | 61 | 125 | 8.3 | 59 | 124 | 8.1 | |
| | 19 | 59 | 120 | 8.0 | 57 | 120 | 7.8 | 55 | 118 | 7.6 | |
| 2300 | 22 | 70 | 132 | 9.4 | 68 | 132 | 9.1 | 65 | 132 | 8.8 | |
| | 21 | 66 | 128 | 8.8 | 63 | 127 | 8.5 | 61 | 127 | 8.3 | |
| | 20 | 61 | 123 | 8.3 | 59 | 122 | 8.0 | 57 | 121 | 7.8 | |
| | 19 | 57 | 118 | 7.8 | 55 | 117 | 7.5 | 53 | 115 | 7.3 | |
| 2200 | 22 | 68 | 129 | 9.1 | 65 | 129 | 8.8 | 63 | 129 | 8.5 | |
| | 21 | 63 | 125 | 8.5 | 61 | 125 | 8.3 | 59 | 124 | 8.0 | |
| | 20 | 59 | 120 | 8.0 | 57 | 120 | 7.8 | 55 | 118 | 7.6 | |
| | 19 | 54 | 115 | 7.5 | 52 | 114 | 7.3 | 51 | 112 | 7.1 | |
| 2100 | 22 | 65 | 127 | 8.7 | 63 | 127 | 8.5 | 60 | 126 | 8.2 | |
| | 21 | 61 | 122 | 8.2 | 59 | 122 | 8.0 | 57 | 121 | 7.7 | |
| | 20 | 56 | 117 | 7.7 | 54 | 116 | 7.5 | 53 | 115 | 7.3 | |
| | 19 | 52 | 112 | 7.3 | 50 | 110 | 7.0 | 49 | 108 | 6.8 | |
| | 18 | 48 | 105 | 6.8 | 46 | 102 | 6.6 | 45 | 99 | 6.4 | |

Figure 5-8. Cruise Performance (Sheet 4 of 6)

- --

20°C ABOVE

STANDARD TEMP

CRUISE PERFORMANCE

PRESSURE ALTITUDE 10,000 FEET

CONDITIONS: 2650 Pounds

Recommended Lean Mixture Cowl Flaps Closed

20°C BELOW

STANDARD TEMP

NOTE
For best fuel economy, operate at the leanest mixture that results in smooth engine operation or at peak EGT if an EGT indicator is installed.

STANDARD

TEMPERATURE

| | | | | -25°C | | | - 5°C | | 15 ⁰ C | | | |
|--|------|----------------------|----------------------|--------------------------|--------------------------|----------------------|--------------------------|--------------------------|----------------------|--------------------------|--------------------------|--|
| | RPM | MP | % BHP | KTAS | GPH | % BHP | KTAS | GPH | % BHP | KTAS | GPH | |
| | 2700 | 20 19 | 72 67 | 136 131 | 9.7 9.0 | 70 65 | 136 131 | 9.3 8.7 | 67 62 | 136 130 | 9.0 8.4 | |
| | 2600 | 20 19 18 | 70 65 60 | 134 129 123 | 9.4 8.8 8.2 | 68 63 58 | 134 128 123 | 9.0 8.5 7.9 | 65 61 56 | 133 128 121 | 8.8 8.2 7.7 | |
| | 2500 | 20 19 18 17 | 68 63 58 54 | 132 127 121 115 | 9.1 8.5 8.0 7.4 | 66 61 56 52 | 132 126 120 113 | 8.8 8.3 7.7 7.2 | 63 59 54 50 | 131 125 119 110 | 8.5 8.0 7.5 7.0 | |
| | 2400 | 20 19 18 17 | 66 61 56 52 | 130 124 119 112 | 8.9 8.3 7.7 7.2 | 63 59 54 50 | 129 124 118 110 | 8.6 8.0 7.5 7.0 | 61 57 52 48 | 129 123 115 107 | 8.3 7.8 7.3 6.8 | |
| | 2300 | 20 19 18 17 | 64 59 54 50 | 127 122 116 109 | 8.6 8.0 7.5 7.0 | 61 57 52 48 | 127 121 114 106 | 8.3 7.8 7.3 6.8 | 59 55 51 46 | 126 119 112 103 | 8.0 7.5 7.1 6.6 | |
| | 2200 | 20 19 18 | 61 57 52 | 125 119 113 | 8.3 7.8 7.3 | 59 55 50 | 124 118 111 | 8.0 7.5 7.0 | 57 53 49 | 123 116 108 | 7.8 7.3 6.9 | |

Figure 5-8. Cruise Performance (Sheet 5 of 6)

57

52

48

121

115

107

7.8

7.3

6.8

20

19

18

59

55

50

122

116

110

8.0

7.5

7.0

2100

7.5

7.1

6.6

119

112

104

55

51

47

PRESSURE ALTITUDE 12,000 FEET

CONDITIONS: 2650 Pounds

Recommended Lean Mixture Cowl Flaps Closed

NOTE

For best fuel economy, operate at the leanest mixture that results in smooth engine operation or at peak EGT if an EGT indicator is installed.

| | | 20°C BELOW STANDARD TEMP -29°C | | | | TANDAF IPERATU -9°C | | 20°C ABOVE STANDARD TEMP 11°C | | | |
|--|------|--------------------------------------|----------------------|--------------------------|--------------------------|---------------------------|--------------------------|-------------------------------------|----------------------|--------------------------|--------------------------|
| | RPM | MP | % BHP | KTAS | GPH | % BHP | KTAS | GPH | % BHP | KTAS | GPH |
| | 2700 | 19 18 | 69 64 | 135 130 | 9.3 8.7 | 67 62 | 135 129 | 9.0 8.4 | 64 60 | 134 128 | 8.7 8.1 |
| | 2600 | 19 18 17 | 67 62 57 | 133 128 121 | 9.0 8.4 7.8 | 65 60 55 | 133 127 120 | 8.7 8.2 7.6 | 63 58 53 | 132 125 117 | 8.5 7.9 7.4 |
| | 2500 | 19 18 17 16 | 65 61 56 51 | 131 126 119 112 | 8.8 8.2 7.6 7.1 | 63 58 54 49 | 131 125 117 108 | 8.5 8.0 7.4 6.9 | 61 56 52 47 | 130 123 115 104 | 8.3 7.7 7.2 6.7 |
| | 2400 | 19 18 17 16 | 63 59 54 49 | 129 123 117 108 | 8.6 7.8 7.4 6.9 | 61 56 52 47 | 128 122 114 105 | 8.3 7.7 7.2 6.7 | 59 54 50 46 | 127 120 111 100 | 8.0 7.5 7.0 6.5 |
| | 2300 | 19 18 17 | 61 57 52 | 126 120 113 | 8.3 7.8 7.2 | 59 54 50 | 125 119 111 | 8.0 7.5 7.0 | 57 53 48 | 124 116 107 | 7.8 7.3 6.8 |
| | 2200 | 19 18 17 | 59 55 50 | 124 118 110 | 8.1 7.5 7.0 | 57 53 48 | 123 115 107 | 7.8 7.3 6.8 | 55 51 46 | 121 112 103 | 7.6 7.1 6.6 |
| | 2100 | 19 18 | 57 52 | 121 114 | 7.8 7.3 | 55 50 | 119 112 | 7.5 7.1 | 53 49 | 117 108 | 7.3 6.9 |

Figure 5-8. Cruise Performance (Sheet 6 of 6)

46

102

6.6

45

98

6.4

17

48

106

6.8

RANGE PROFILE 45 MINUTES RESERVE 62 GALLONS USABLE FUEL

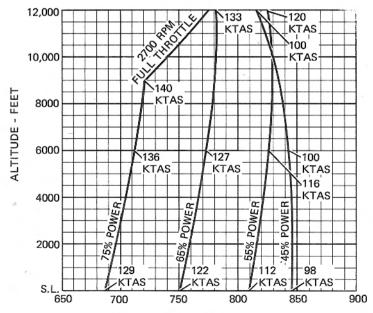
CONDITIONS: 2650 Pounds

Recommended Lean Mixture for Cruise Standard Temperature

Zero Wind:

NOTE:

This chart allows for the fuel used for engine start, taxi, takeoff and climb, and the distance during a normal climb up to 8000 feet and maximum climb above 8000 feet.



RANGE - NAUTICAL MILES

Figure 5-9. Range Profile (Sheet 2 of 2)

ENDURANCE PROFILE 45 MINUTES RESERVE 44 GALLONS USABLE FUEL

CONDITIONS: 2650 Pounds

Recommended Lean Mixture for Cruise Standard Temperature

NOTE:

This chart allows for the fuel used for engine start, taxi, takeoff and climb, and the time during a normal climb up to 8000 feet and maximum climb above 8000 feet.

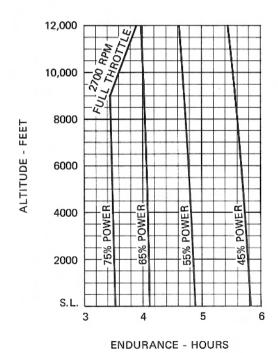


Figure 5-10. Endurance Profile (Sheet 1 of 2)

ENDURANCE PROFILE 45 MINUTES RESERVE

62 GALLONS USABLE FUEL

CONDITIONS:

2650 Pounds

Recommended Lean Mixture for Cruise

Standard Temperature

NOTE:

This chart allows for the fuel used for engine start, taxi, takeoff and climb, and the time during a normal climb up to 8000 feet and maximum climb above 8000 feet,

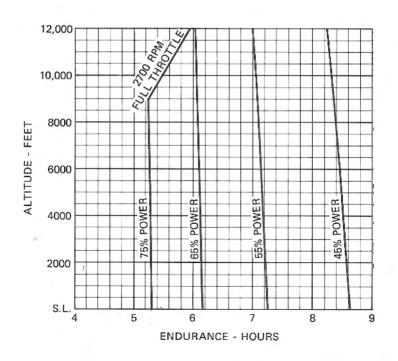


Figure 5-10. Endurance Profile (Sheet 2 of 2)

LANDING DISTANCE

SHORT FIELD

CONDITIONS: Flaps 300

Power Off Maximum Braking

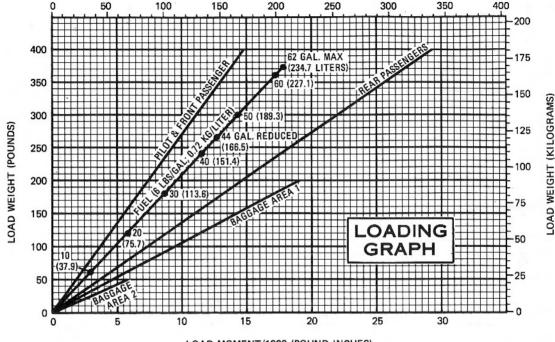
Paved, Level, Dry Runway Zero Wind

NOTES:

- 1. Short field technique as specified in Section 4.
- Decrease distances 10% for each 9 knots headwind. For operation with tailwinds up to 10 knots, increase distances by 10% for each 2 knots.
- 3. For operation on a dry, grass runway, increase distances by 40% of the "ground roll" figure.
- If a landing with flaps up is necessary, increase the approach speed by 9 KIAS and allow for 35% longer distances.

| WEIGHT LBS | SPEED AT 50 FT KIAS | PRESS ALT FT | 0°C | | 10 ^o C | | 20°C | | 30°C | | 40°C | |
|---------------|------------------------------|---|---|--|---|--|--|--|---|--|---|--|
| | | | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | The Control of the Co | TOTAL TO CLEAR 50 FT OBS | GRND ROLL | TOTAL TO CLEAR 50 FT OBS | | TOTAL TO CLEAR 50 FT OBS |
| 2650 | 63 | S. L. 1000 2000 3000 4000 5000 6000 7000 8000 | 590 615 635 660 685 710 740 770 800 | 1290 1325 1355 1395 1430 1470 1515 1560 1605 | 615 635 660 685 710 740 765 795 825 | 1325 1355 1395 1430 1470 1515 1555 1600 1645 | 635 660 685 710 735 765 795 825 855 | 1355 1395 1430 1470 1510 1550 1595 1645 1690 | 660 680 705 735 760 790 820 850 885 | 1390 1425 1465 1505 1545 1590 1635 1685 1735 | 680 705 730 760 785 815 850 880 915 | 1425 1460 1500 1545 1585 1630 1680 1725 1780 |





LOAD MOMENT/1000 (POUND-INCHES)

NOTES: Line representing adjustable seats shows the pilot or passenger center of gravity on adjustable seats positioned for an average occupant. Refer to the Loading Arrangements diagram for forward and aft limits of occupant C.G. range.

Figure 6-6. Loading Graph

LOADED AIRPLANE WEIGHT (KILOGRAMS)

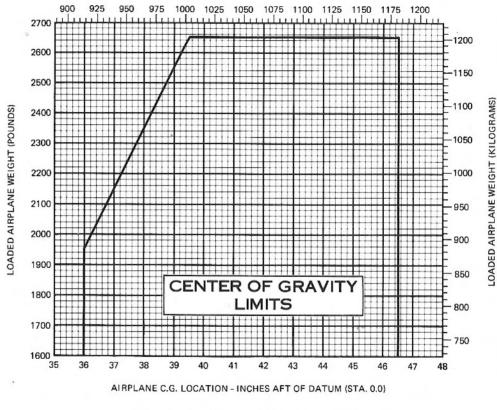
-1150

- 750

LOADED AIRPLANE MOMENT/1000 (KILOGRAM-MILLIMETERS) CENTER OF GRAVITY LOADED AIRPLANE WEIGHT (POUNDS)

Figure 6-7. Center of Gravity Moment Envelope

LOADED AIRPLANE MOMENT/1000 (POUND-INCHES)



AIRPLANE C.G. LOCATION - MILLIMETERS AFT OF DATUM (STA. 0.0)

Figure 6-8. Center of Gravity Limits

LOADING ARRANGEMENTS

- *Pilot or passenger center of gravity on adjustable seats positioned for average occupant. Numbers in parentheses indicate forward and aft limits of occupant center of gravity range.
- **Arm measured to the center of the areas shown.
- NOTES: 1. The usable fuel C.G. arm is located at station 48.0.
 - The rear cabin wall (approximate station 108) or aft baggage wall (approximate station 124) can be used as convenient interior reference points for determining the location of baggage area fuselage stations.

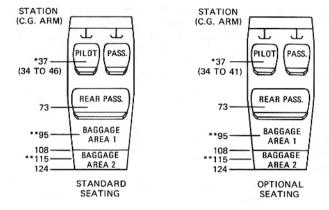


Figure 6-3. Loading Arrangements